EOS-AM2 MISSION

PRESENTATION TO SCIENCE
WORKING GROUP, AM PROJECT
GSFC
APRIL 4, 1997

MISSION ASSUMPTIONS

- LANDSAT AND EOS-AM MISSIONS MERGE AFTER LANDSAT-7 AND EOS-AM1
 - THE FIRST SPACECRAFT IS EOS-AM2
- LAUNCH DATE IS JUNE 2004
- MEASUREMENT COMPLEMENT ENCOMPASSES THE MEASUREMENTS OF MODIS, MISR, ETM+, CERES, EOSP
- SPACECRAFT IS COMPATIBLE WITH GROUND STATIONS AND WITH THE EOS GROUND SYSTEM
- ACCOMMODATE LANDSAT INTERNATIONAL GROUND STATIONS

MISSION OBJECTIVES

- MEET REQUIREMENTS OF MTPE SCIENCE RESEARCH PLAN AND LANDSAT FOLLOW-ON (P.L. 102-555)
 - MEASUREMENT CONTINUITY
 - CO-REGISTRATION
 - GEOLOCATION
 - INTERNATIONAL LANDSAT GROUND STATIONS
- INCORPORATE LESSONS LEARNED FROM EOS-AM1 AND LANDSAT-7
 - CALIBRATION MANEUVERS
- INCORPORATE ADVANCED TECHNOLOGIES FOR IMPROVED SYSTEM PERFORMANCE

MISSION DESIGN OPTIONS

- FLY COPIES OF EXISTING INSTRUMENTS AND SPACECRAFT
 - MINIMAL DEVELOPMENT
 - LOWEST RISK
 - STATIC TECHNOLOGY
 - EXISTING INFRASTRUCTURE (ALGORITHMS, SPACECRAFT)
- FLY EVOLUTIONARY INSTRUMENTS AND SPACECRAFT
 - MODIFY BASIC DESIGN
 - REPLACE OBSOLETE ELEMENTS
 - INFRASTRUCTURE MAY REQUIRE MODIFICATIONS
- NEW DESIGN
 - NEED DEVELOPMENT
 - HIGHER RISK
 - DYNAMIC TECHNOLOGY
 - INFRASTRUCTURE MUST BE MODIFIED
 - COST TRADE MUST BE UNDERSTOOD
- RELY ON OTHER PROVIDERS (COMMERCIAL, INTERNATIONAL, OTHER AGENCY)
 - ACCEPT WHAT IS PROVIDED

SUMMARY OF OPTIONS

OPTION	1	2	3	4	5	6
	BASELINE	ATLAS	LIGHT	INTEGRATED	FORMATION FLYING	DOCKING
DESCRIPTION	8/94 BASELINE	UTILIZE AM-1 DESIGN	MODIFY INSTRUMENTS	INTEGRATED PAYLOAD ARCHITECTURE	FORMATION FLYING OF PAYLOAD	ORBITAL ASSEMBLY OF PAYLOAD COMPLEMENT
INSTRUMENTS	MOOIS MISR ETM+ OR NEW TECHNOLOGY CERES EOSP	MODIS MISR ETM+ OR NEW TECHNOLOGY CERES EOSP	MODIS / LIGHT MISH / LIGHT ETM- / LIGHT OR NEW TECHNOLOGY CERES EOSP	MODIS BANDS + OPTICS + A/D ONLY MISR BANDS + OPTICS + A/D ONLY ETM+ BANDS + OPTICS + A/D ONLY CERES EOSP	MODIS BANDS + OPTICS + A/D ONLY MISR BANDS + OPTICS + A/D ONLY ETM+ BANDS + OPTICS + A/D ONLY CERES EOSP	MODIS BANDS + OPTICS + A/D ONLY MISR BANDS + OPTICS + A/D ONLY ETM+ BANDS + OPTICS + A/D ONLY CERES EOSP
NOTES ON INSTRUMENTS	INSTRUMENTS AS CURRENTLY EXIST (OR NEW TECHNOLOGY FOR LANOSAT MEASUREMENTS).	INSTRUMENTS AS CURRENTLY EXIST (OR NEW TECHNOLOGY FOR LANOSAT MEASUREMENTS).	'LIGHT' INSTRUMENTS ARE SEPARATE INSTRUMENTS DERIVING FROM EXISTING INSTRUMENTS, PROBABLY FROM CURRENT DEVELOPERS, DELETE MOST DATA HANDLING AND CONTROL ELECTRONICS	IMAGERS ARE INTEGRATED SET, POSSIBLY FROM A SINGLE DEVELOPER.	IMAGING SYSTEMS WITHIN A SINGLE GROUP OF BANDS (MODIS, MISR, ETM+) ARE STAND ALONE SYSTEMS.	IMAGING SYSTEMS WITHIN A SINGLE GROUP OF BANDS (MODIS, MISR, ETM4) ARE STAND ALONE SYSTEMS
SPACECRAFT	COMMON	AM1 OPTION B	SIZE IN STUDY	SIZE IN STUDY	SIZE 5 SPACECRAFT IN STUDY	SIZE 5 SPACECRAFT IN STUDY
NOTES ON SPACECRAFT	MODIFICATIONS REQUIRED FOR REVISED INSTRUMENT PAYLOAD	AM1 OPTION B DOES NOT INCLUDE INSTRUMENT INTEGRATION.	SPACECRAFT INCORPORATES DATA BUFFERS, COMPUTING, AND POWER CONDITIONING REQUIRED BY 'LIGHT' INSTRUMENTS.	SPACECRAFT INCORPORATES DATA BUFFERS, COMPUTING, AND POWER CONDITIONING REQUIRED BY INTEGRATED IMAGER SET. DATA SYSTEM IS PART OF SPACECRAFT CONTRACT, OR GFE.	5 STAND ALONE SPACECRAFT, ONE INSTRUMENT PER SPACECRAFT, FLOWN AND MAINTAINED IN PROXIMITY.	ONE SPACECRAFT MAKES CORE SUPPORT SERVICES SYSTEM PLUS ONE INSTRUMENT. FOUR ADDITIONAL SPACECRAFT, ONE INSTRUMENT PER SPACECRAFT, ALL DOCKED TO CORE SYSTEM.
LAUNCH VEHICLE	DELTA	ATLAS IIAS	DETERMINE IN STUDY, PROBABLY DELTA	DETERMINE IN STUDY, PROBABLY DELTA LITE OR SMALLER	DETERMINE IN STUDY, PROBABLY: 4 - 5 TAURUS OR PEGASUS 3 - 5 LLV	DETERMINE IN STUDY, PROBABLY: 4 - 5 TAURUS OR PEGASUS 3 - 5 LLV
NUMBER OF LAUNCHES	1	1	1	1	4-5 TAURUS OR PEGASUS 3-5 LLV	4 - 5 TAURUS OR PEGASUS 3 - 5 LLV
OPTIONAL SENSORS	NEW TECHNOLOGY SENSOR (OPTICAL, SAR).	NEW TECHNOLOGY SENSOR (OPTICAL, SAR).	NEW TECHNOLOGY SENSOR (OPTICAL, SAR).	NEW TECHNOLOGY SENSOR (OPTICAL, SAR).	NEW TECHNOLOGY SENSOR (OPTICAL, SAR).	NEW TECHNOLOGY SENSOR (OPTICAL, SAR).
LAUNCH COSTS	\$44 - 51M	\$105 - 115M	\$44 - 51M	\$25 - 30M	\$65 - 115M	\$85 - 115M
Ps FOR PUTTING COMPLETE PAYLOAD ON ORBIT	0.99	0.63	0.99	TBD - 0.99	0.03 - 0.5 (PEGASUS)	0.03 - 0.5 (PEGASUS)
CONCERNS	VIABILITY OF COMMON SPACECRAFT USAGE POST X-1 PHASE OF EOS 2 FIELD OF VIEW OBSTRUCTIONS 3. POINTING PERFORMANCE 4. DATA RATE PERFORMANCE 5. DATA CAPACITY PERFORMANCE	1, VIABILITY OF CONTINUED USE OF ATLAS 2, AVAILABILITY OF PARTS IN 1998 TO 2004 TIME FRAME 3, AVAILABILITY OF TORSS	2. POINTING PERFORMANCE	NEW TECHNOLOGY MUST BE DEMONSTRATED COMMITMENT OF SCIENCE COMMUNITY TO INTEGRATE FUNCTIONS (DELETE FROM SINGLE INSTRUMENTS)	1. REDUCED PROBABILITY OF SUCCESSFUL CONSTELLATION 2. RELIABILITY OF PEGASUS 3. INCREASED LAUNCH COSTS 4. INTEGRATION OPPORTUNITIES LIMITED 5. POINTING CONCERNS 6. LAUNCH WINDOW CONSTRAINTS 7. COMMUNICATIONS CONCERNS 8. GROUND OPS COMPLICATED 9. DATA PROCESSING OR SCIENCE CONCERNS	REDUCED PROBABILITY OF SUCCESSFUL CONSTELLATION RELIABILITY OF PEGASUS INCREASED LAUNCH COSTS INTEGRATION OPPORTUNITIES LIMITED S. POINTING CONCERNS LAUNCH WINDOW CONSTRAINTS
DEMO REQUIRED	NO	NO	NO	YES - SATISFY USER COMMUNITY OF FULL CONTINUITY	NO - FOR FORMATION FLYING (ALREADY DONE) YES - FOR INSTRUMENT SUITE CHANGES	ROUTINELY DONE BY RUSSIANS

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TECHNOLOGY CONSIDERATIONS FOR THE AM-2 ERA

INSTRUMENTS

ADVANCED GLOBAL IMAGER

OBJECTIVES SATISFY NOAA OPERATIONAL NEEDS

IMPLEMENT IMPROVED UNDERSTANDING FROM MODIS DEVELOPMENT AND OPS

TRADES SCANNER OR PUSHBROOM

SINGLE OR MULTIPLE IMAGER UNITS REFLECTIVE OR REFRACTIVE OPTICS MULTISPECTRAL OR HYPERSPECTRAL

CALIBRATION (ON-BOARD, GROUND-BASED ACTIVE, MANEUVERS)

GEOLOCATION (REALTIME OR POST-PROCESSED) . CO-REGISTRATION (REALTIME OR POST-PROCESSED)

TECHNOLOGY QWIP OR BOLOMETER ARRAY DETECTORS (TIR)

INGAS OR SI DETECTORS (SWIR)

STIRLING CYCLE, PULSE TUBE, OR PASSIVE COOLERS

DATA COMPRESSION

DRIVERS STABILITY, POINTING KNOWLEDGE, COOLING, DATA RATE

HIGH RESOLUTION IMAGER

OBJECTIVES SATISFY REQUIREMENTS OF P.L. 102-555

IMPROVEMENT OVER ETM+ AND PROPOSED COMMERCIAL.

IMPLEMENT IMPROVED UNDERSTANDING FROM ETM + EXPERIENCE

TRADES SCANNER OR PUSHBROOM OR STARING ARRAY

SINGLE OR MULTIPLE IMAGER UNITS REFLECTIVE OR REFRACTIVE OPTICS

THERMAL BAND ACTIVE OR PASSIVE

MULTISPECRAL OR HYPERSPECTRAL

CALIBRATION (ON-BOARD, GROUND-BASED, MANEUVERS)

GEOLOCATION (REALTIME OR POST-PROCESSED)
CO-REGISTRATION (REALTIME OR POST-PROCESSED)

TECHNOLOGY INGAS OR SI DETECTORS (SWIR)

ACTIVE OR PASSIVE PIXEL ARRAYS

ACTIVE OR PASSIVE COOLING

DISCRETE OR WEDGE FILTER, OR GRATING

DRIVERS SPACECRAFT STABILITY, COOLING, POINTING KNOWLEDGE, DATA RATE

COMMUNICATIONS (REGULAR DOWNLINK AND INTERNATIONAL USERS)

ADVANCED MISR

OBJECTIVES BRDF (BI-DIRECTIONAL REFLECTANCE DISTRIBUTION FUNCTION) AND AFROSOLS

TRADES NINE OR FIVE CAMERAS

REFRACTIVE OR REFLECTIVE OPTICS PASSIVE OR ACTIVE PIXEL ARRAYS FOUR BANDS OR EIGHT BANDS

CALIBRATION (ON-BOARD, GROUND-BASED, MANEUVERS)

GEOLOCATION (REALTIME OR POST-PROCESSED)
CO-REGISTRATION (REALTIME OR POST-PROCESSED)

TECHNOLOGY ACTIVE PIXEL ARRAYS

DRIVERS STABILITY, POINTING KNOWLEDGE, COOLING, DATA RATE

CERES

OBJECTIVES IMPLEMENT IMPROVED UNDERSTANDING IDENTIFIED DURING CERES DEV

TRADES COPY OR NEW DEVELOPMENT

DRIVERS DISTURBANCE (REACTION TORQUE)

EOSP

OBJECTIVES IMPLEMENT NEW SCIENCE AND NEW MEASUREMENTS

TRADES ALONG-TRACK OR CROSS-TRACK SCANNER

NUMBER OF BANDS

FILTER TECHNIQUES (DISCRETE OR PEM)

SPACECRAFT

OBJECTIVES

AUTONOMOUS OPERATIONS

DISTRIBUTE DATA TO USERS

REDUCE COST AND DEVELOPMENT TIME

TRADES

NUMBER OF SPACECRAFT (ONE, THREE, FIVE, SEVEN)

DOWNLINK (PHASED ARRAY KU, KA, OPTICAL)

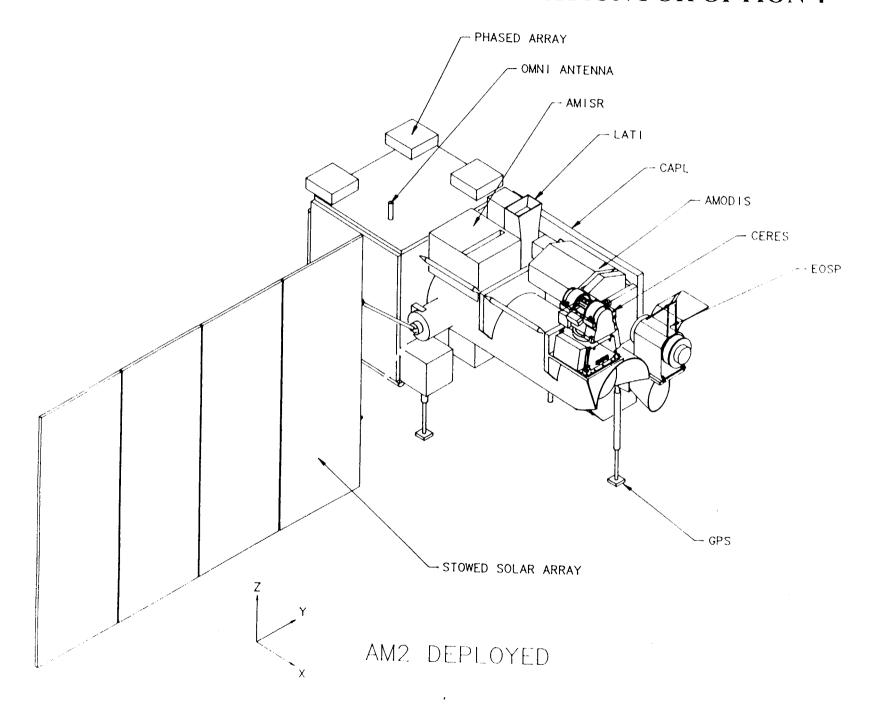
STAR CAMERAS OR STAR TRACKERS

INTEGRATED OR DISTRIBUTED DATA SYSTEM

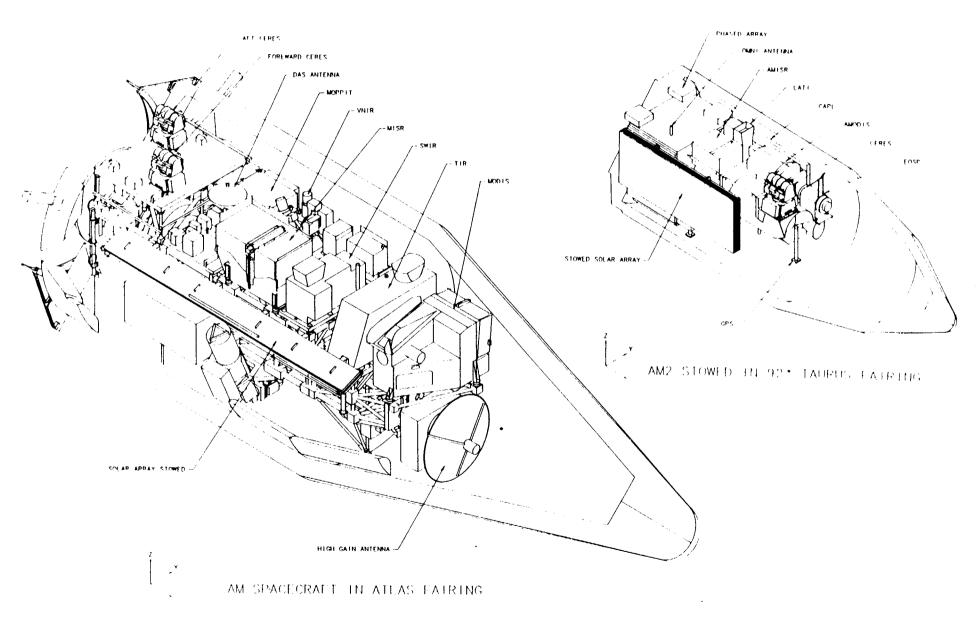
ACTIVE DISTURBANCE SUPPRESSION

FORMATION FLYING CONTROL (ACTIVE OR PASSIVE)

EXAMPLE INSTRUMENT ACCOMMODATION FOR OPTION 4

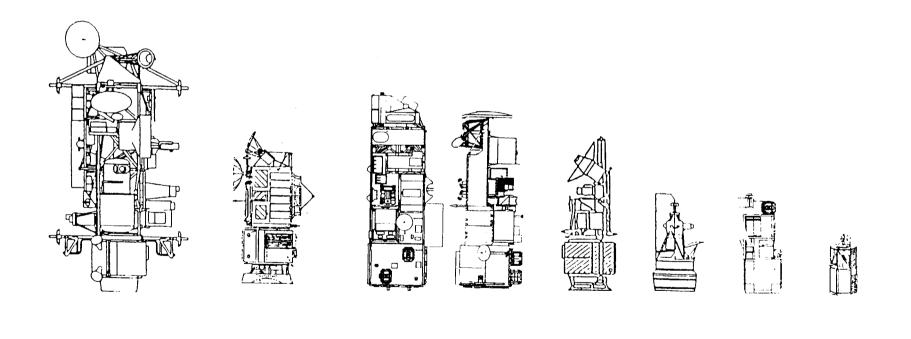


COMPARISON OF OPTION 4 EOS AM-2 CONCEPT TO EOS AM-1



OPTIONS 4 AND 5 YIELD CONSIDERABLY SMALLER SPACECRAFT

Spacecraft Comparisons



EOS-PM1

LANDSAT-4 NIMBUS-7

EOS-AM2 SEASTAR

EOS-AM1

UARS

TOPEX/

POSEIDEN

EXAMPLE CONSTELLATION NAVIGATION AND GEOLOCATION REQUIREMENTS

EOS-AM2 SPACECRAFT TRIAD (0.1 pixel coregistration criterion)

	Simultaneity (min)		Pixel Geolocation Knowledge (m, 3σ) AT/XT	S/C Position Knowlege (m, 3 ₀) AT-XT-R	Pointing Knowledge (arcsec, 3σ)	Pointing Accuracy (arcsec, 3ਰ)
MODIS-CERES	} } 3.5	}	71/71	20-20-20 Geode 5-5-5 WAAS	20-20-20 21-21-21	360-360-360 ²
MISR-EOSP	}	}1 cloud studies }15 aerosol corr.	71/71	20-20-20 Geode 5-5-5 WAAS	20-20-20 21-21-21	1373-1373-1373 ²
LATI		}	71/71	20-20-20 Geode 5-5-5 WAAS	20-20-20 21-21-21	180-180-180

EXAMPLE RELATED INTERNATIONAL INSTRUMENTS

Instrument Name	Agency	Mission ¹	Mass (kg)	Power (W)	D. Rate (Mbps) F	AM2 Replacement
PRISM (Hyperspectral & TIR 50m Imager)	ESA	EE(> '02)	400	400	260	LATI?
MERIS (Land or Ocean Color)	ESA	EW(> '02)	180	230	24-1.6 MC	DDIS(VNIR)?
AATSR	ESA/UK	EW(> '02) METOP-1('01)	99	111	0.7 M	IODIS(SST's)
SCARAB	ESA/CNES	EE(> '02) METOP-1('01)	42 ²	40-60 ²	0.001 ²	CERES
POLDER 3	CNES	ADEOS-3('05)? EE(> '02)?	32	50	0.882	EOSP
GLI 2	NASDA	ADEOS-3('05)	~350 ³	~400 ³	203	MODIS
AVNIR-3	NASDA	ALOS-2('07)	250	230	240	LATI?
LISS-III, IV	ISRO	IRS-2A,B (> '00)?	81	85	37.7	LATI?

NOTES.

^{1.} EE - Earth Explorer, EW - Earth Watch, ALOS - Advanced Land Observing Satellite, IRS - Indian Remote Sensing Satellite.

^{2.} Accommodation resources for Meteor/3 version.

^{3.} Accommodation resources based on Phase B results.

CONCLUSION

- CONTINUING TO IDENTIFY AND EVALUATE ADVANCED TECHNOLOGIES AND SYSTEM ARCHITECTURES FOR MISSION
 - DETERMINED THAT SCIENCE REQUIREMENTS CAN BE MET THROUGH A COMBINATION OF ADVANCED AND ALTERNATIVE TECHNOLOGES
- EVALUATING FEASIBILITY OF MULTIPLE SPACECRAFT APPROACH
 - SCIENCE
 - TECHNOLOGY
 - COST
- INVESTIGATING EXPANDED (E.G. HYPERSPECTRAL) AND COMPLEMENTARY E.G. ACTIVE) MEASUREMENTS MADE POSSIBLE THROUGH ADVANCED TECHNOLOGY